



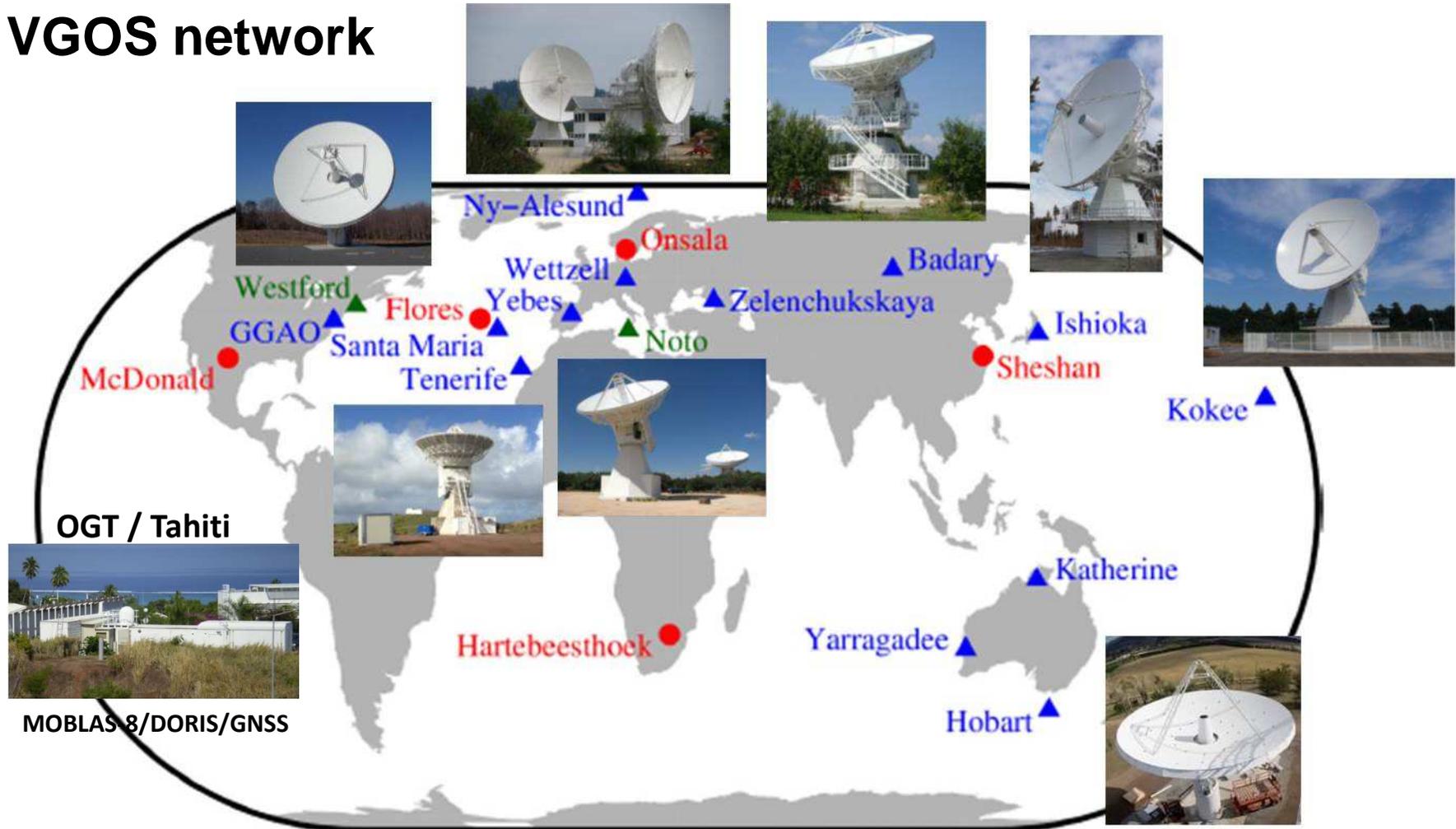
Plan for a VLBI antenna in Tahiti from 2018

Richard BIANCALE, Centre National d'Etudes Spatiales, France,

Jean-Pierre BARRIOT, Université de la Polynésie Française, French Polynesia

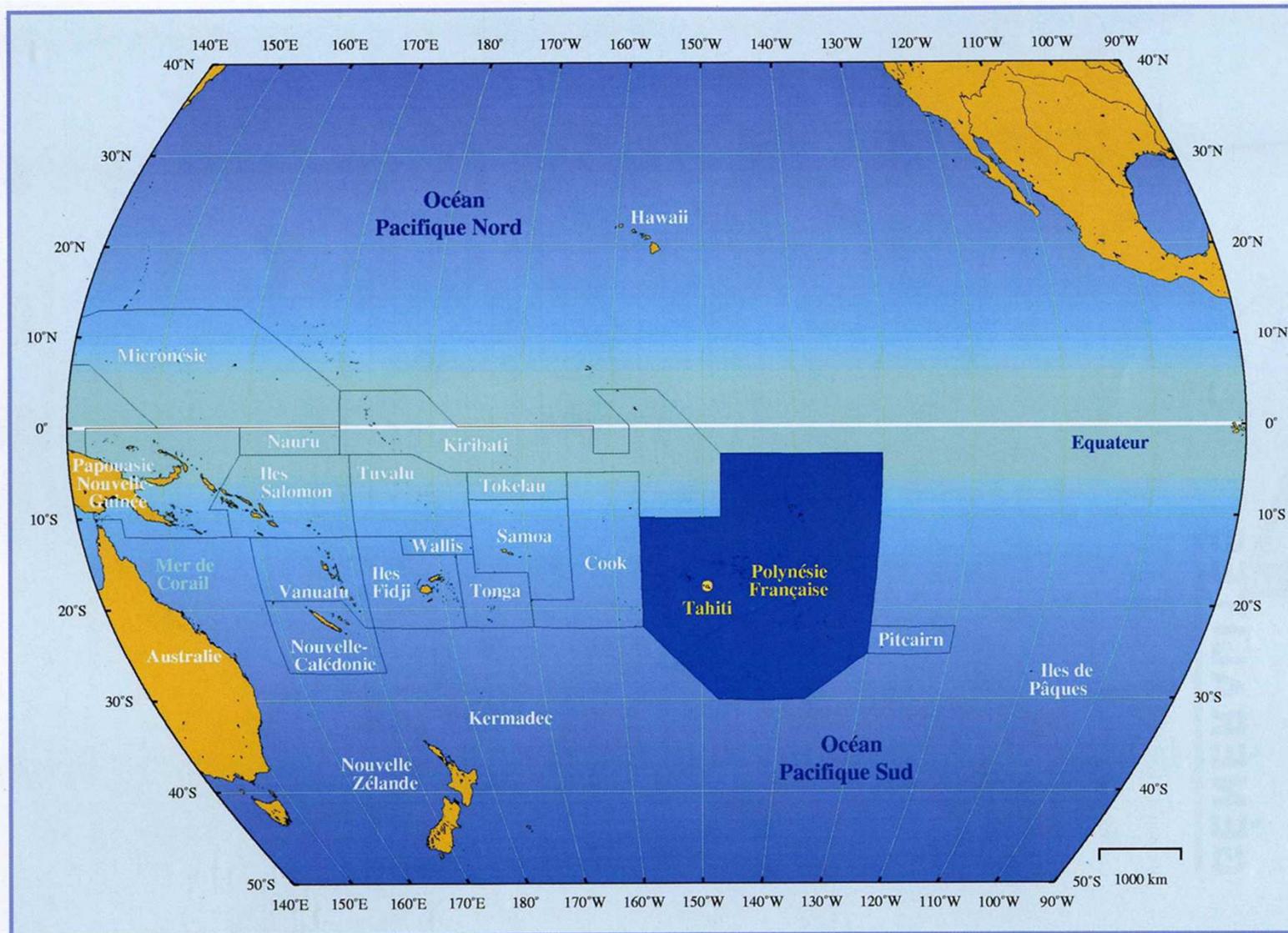


VGOS network



VGOS progress:

- ▲ hardware work in progress
- funding approved
- ▲ legacy upgrade in progress



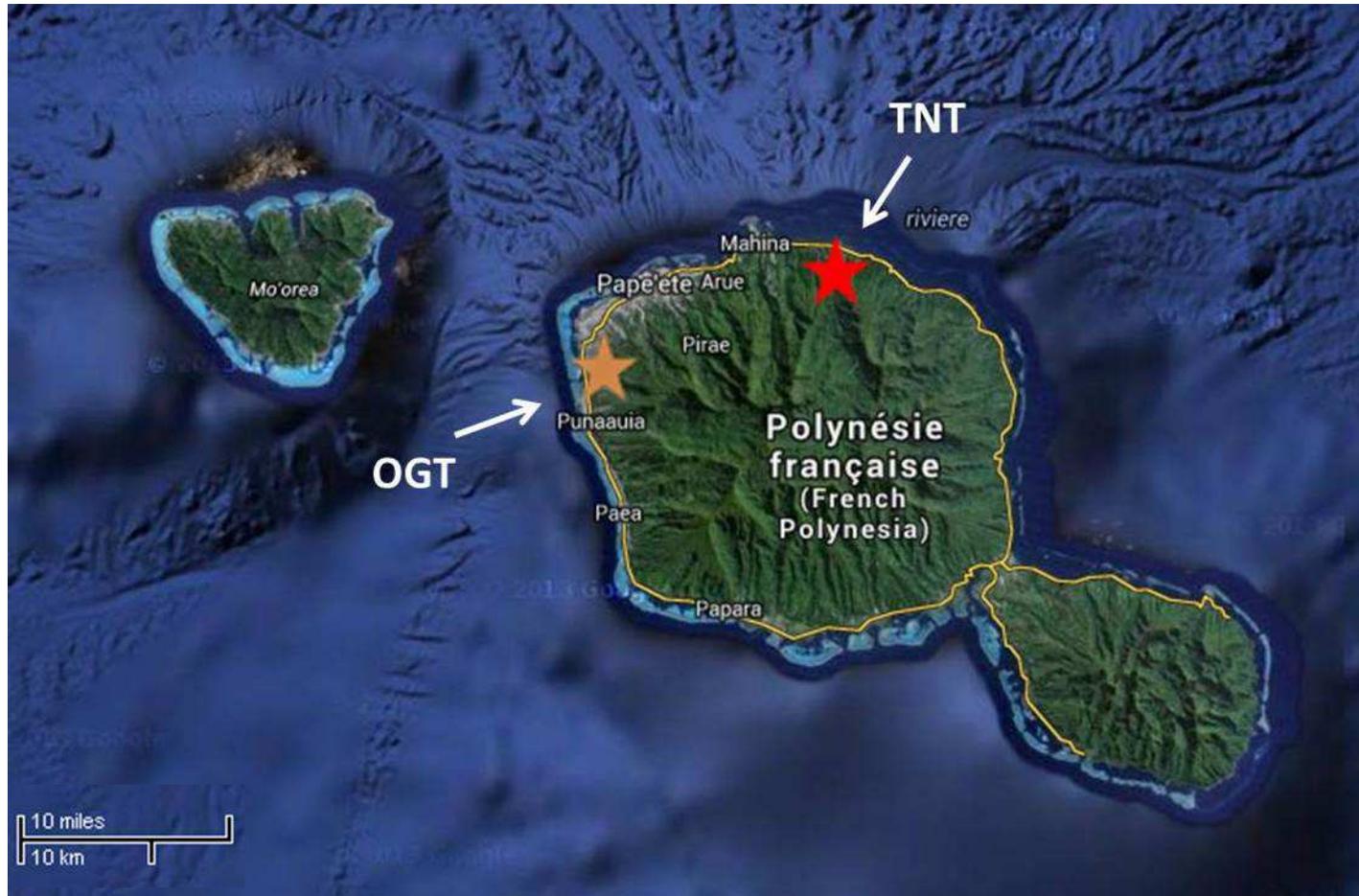
Météo France, 2004

French territory

Maritime area: $5.5 \cdot 10^6 \text{ km}^2$ (half so large as Europe or USA)

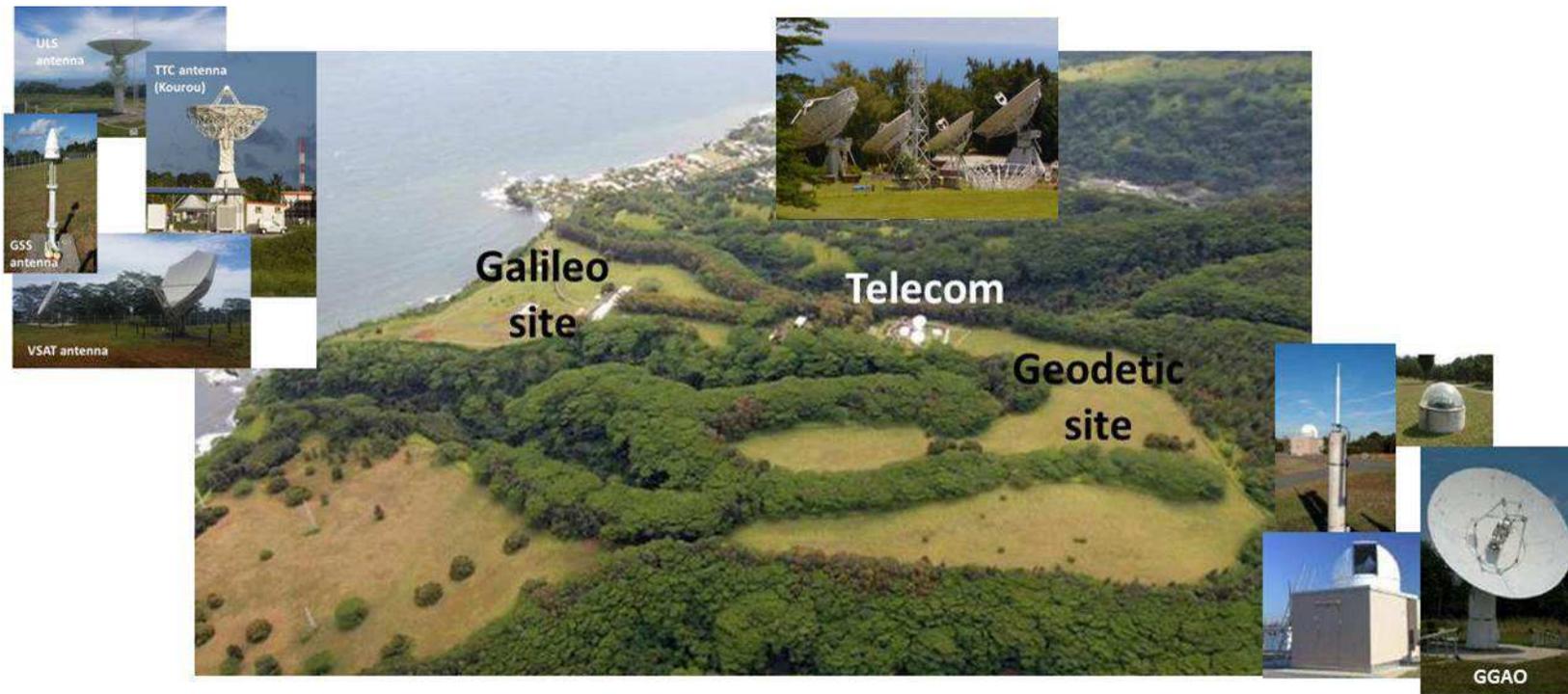
Land area: $3\,500 \text{ km}^2$ (118 islands and atolls, 76 inhabited, 280 000 inhabitants)

The Tahiti Nui Telecom (TNT) site



The Tahiti Nui Telecom site

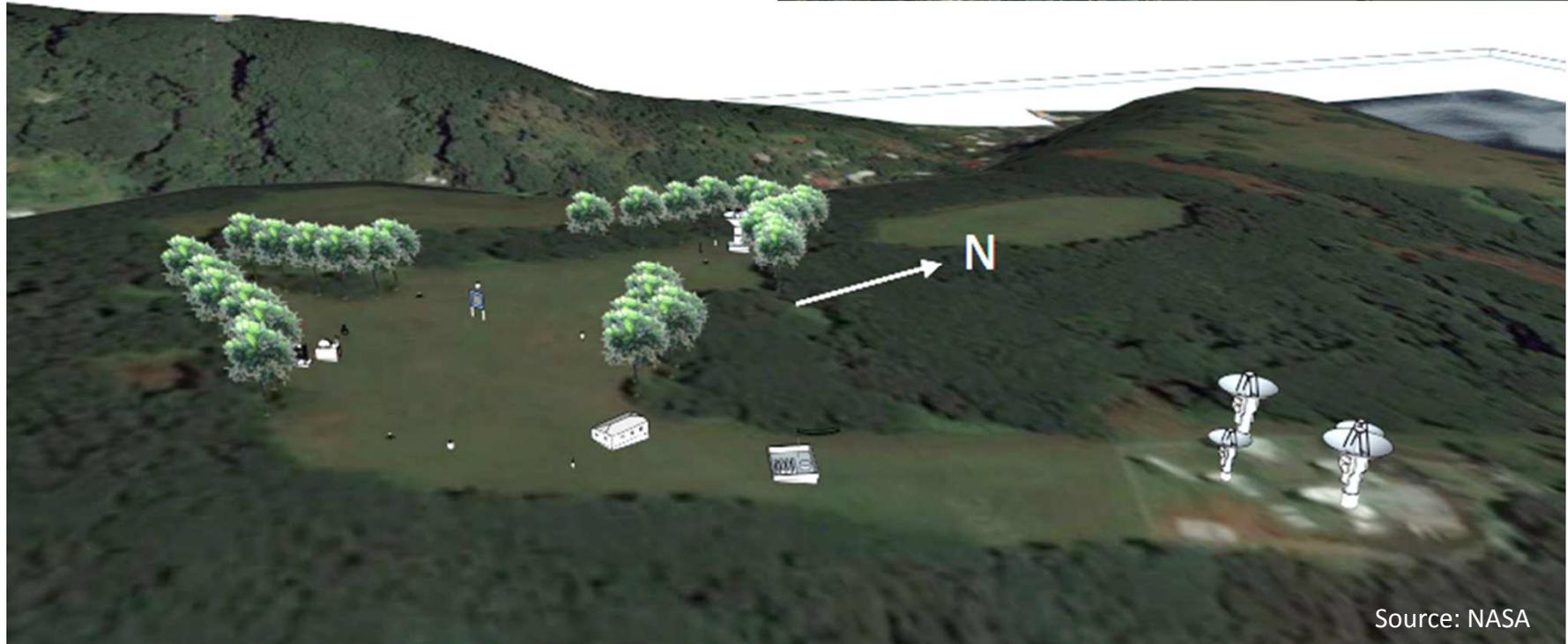
- Tahiti Nui Telecom (TNT) expressed its agreement in principle for hosting instruments of space geodesy on its ground in Papenoo for a 30 year period at least.



- The site of Papenoo benefits from a recognized natural and physical protection. It already hosts the chief station of satellite reception in Tahiti (4.2 and 6 GHz), a ground station of the Galileo system (1.5 and 5.2 GHz) as well as the Tahitian extremity of the international Honotua submarine fiber from the Hawaiian Islands.

NASA layout proposal

- ~3 ha site in the South clearing of the Atohei plateau belonging to TNT (72 ha)
- 17.5178°S, 149.4370°W, 200m alt.
- NASA will expertise the site in April 2016 (RFI tests)



VLBI simulations

A VLBI antenna in Tahiti would complete profitably the GGOS network . The plan is to rely on the development and installation of a new NASA VGOS antenna in 2018 in the framework of the CNES-NASA cooperation on space geodesy activities (NASA-CNES Implementing Arrangement, 2014).

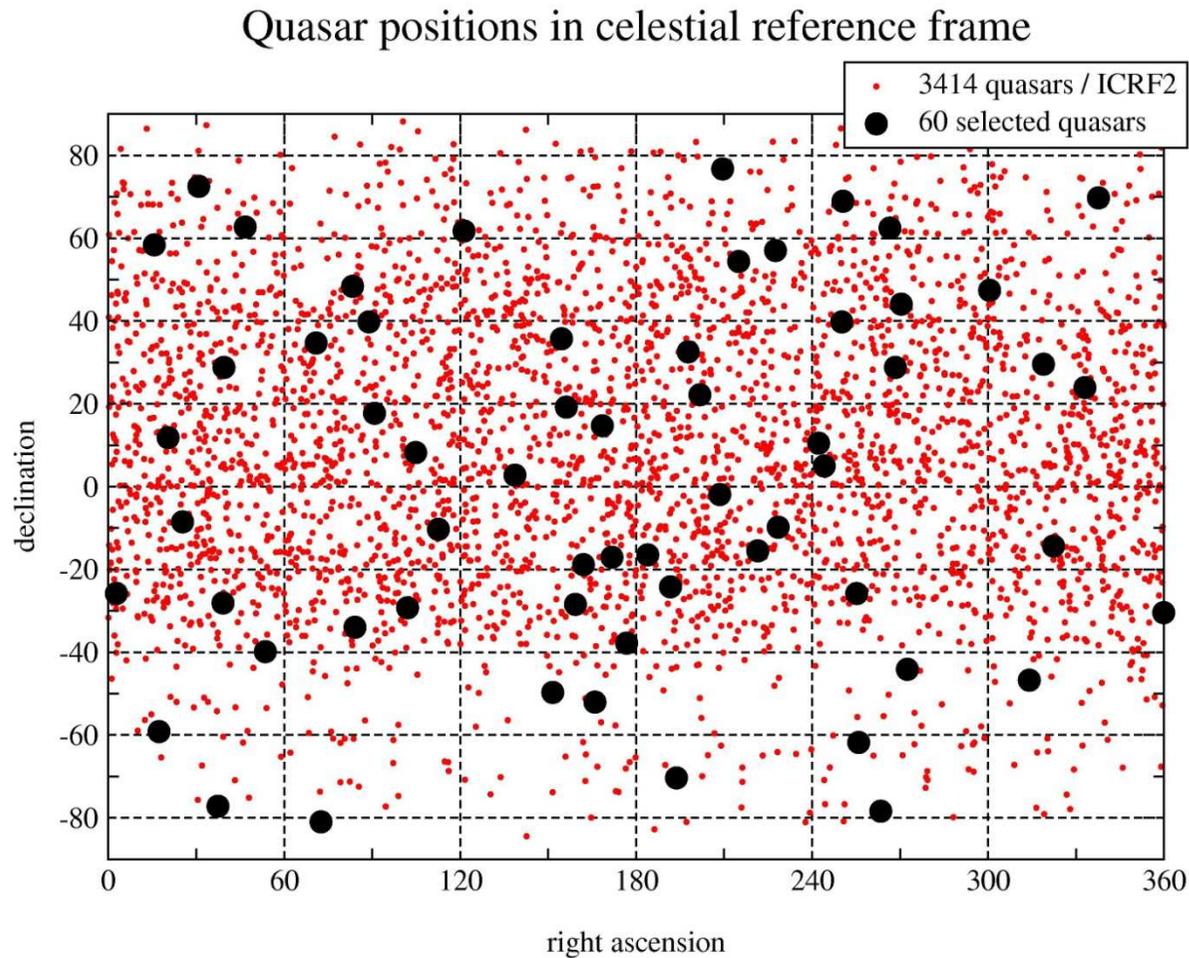
Some simulation studies were already performed (eg from D. MacMillan, NASA/GSFC, 2010):

- *adding Tahiti to a 8-station network (Hobart, Kokee, Canary Isl., NyAlesund, Tsukuba, GGAO, Wettzell, Badary) improves EOP precision by 25%,*
- *taking into account a set of 15-station globally distributed network, adding Tahiti improves EOP precision by 13%.*

The CNES/GRGS GINS software is able to simulate all kinds of space geodesy measurements. This feature was used to estimate the impact of an additional VLBI antenna in Tahiti taking into account a limited set of 12 VLBI antennas well distributed all over the world with a set of 60 quasar sources.

Set of quasars for simulations

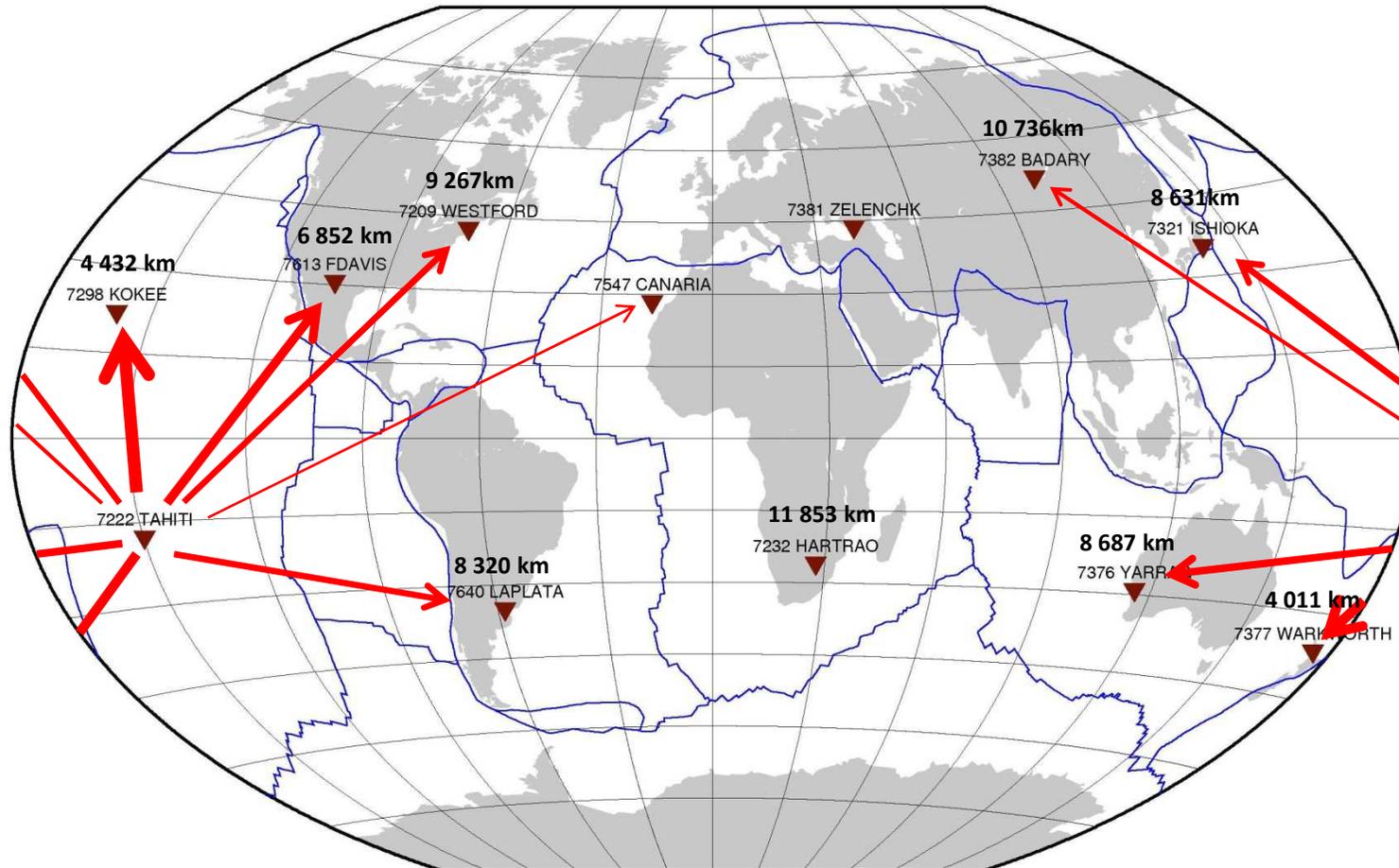
From 295 ICRF2 quasars used for the celestial reference frame orientation we selected a set of 60 quasars equally distributed in right ascension and declination



VLBI station network

Example of simulated data over one week from 12 stations and 60 quasars (~28 000 obs., ~4 800 obs. from Tahiti)

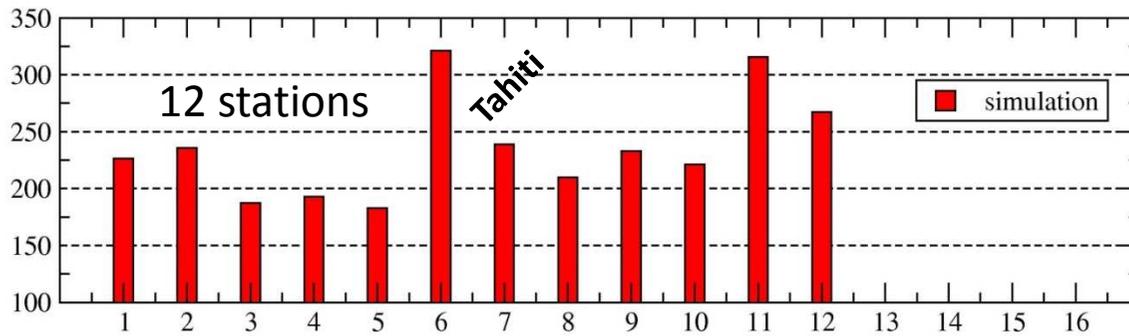
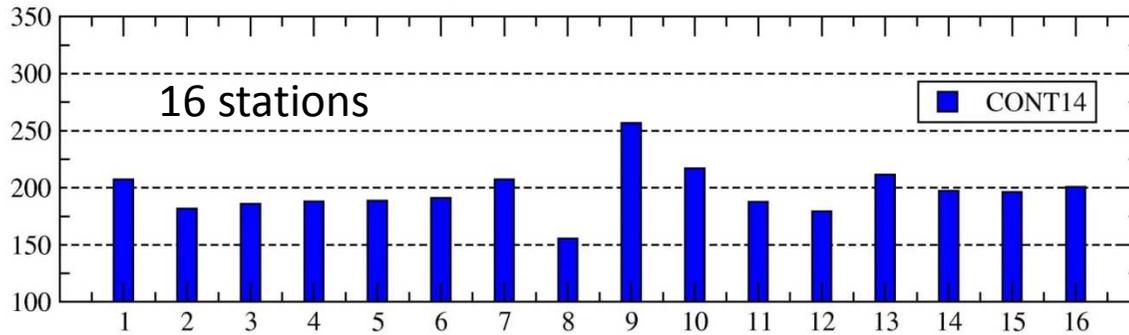
bases with Tahiti per week:		
752 obs.	TAHITI - KOKEE	4431 km
715 obs.	TAHITI - WARKWORT	4011 km
667 obs.	TAHITI - FDAVIS	6851 km
472 obs.	TAHITI - YARRAG	8687 km
392 obs.	TAHITI - WESTFORD	9267 km
370 obs.	TAHITI - LAPLATA	8319 km
392 obs.	TAHITI - ISHIOKA	8630 km
84 obs.	TAHITI - BADARY	10736 km
15 obs.	TAHITI - CANARIA	11817 km



Simulation vs. CONT'14 data

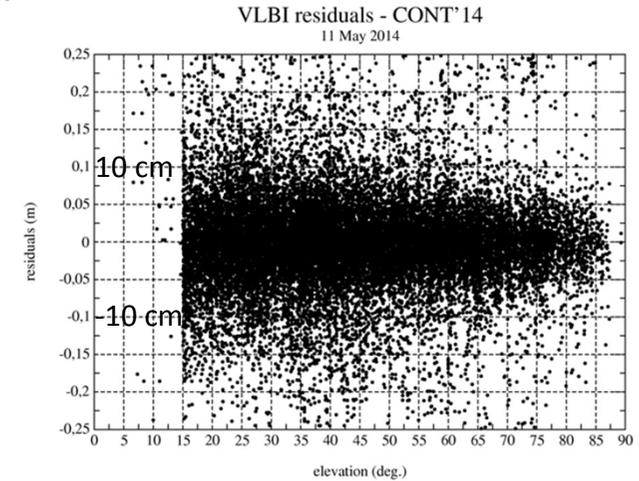
In order to be most realistic we considered equivalent observation density (one observation every ~200s in average per station) and simulated model errors giving similar results as for the CONT'14 campaign.

Mean scan interval (in sec.) per station over one week

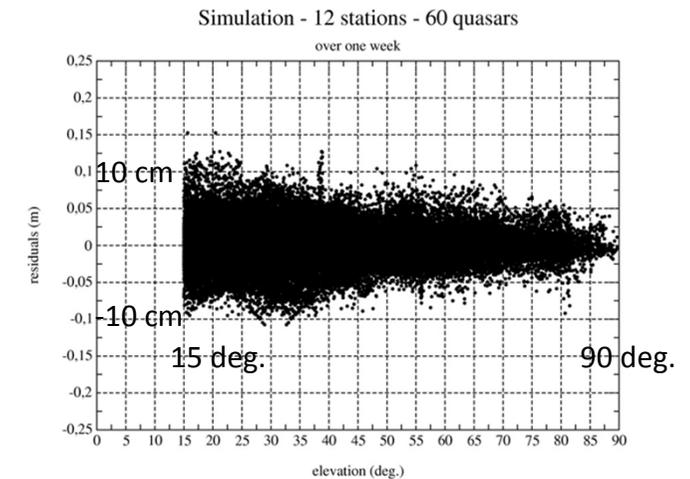


Observations simulated every 90 s → 28 000 observations per week

CONT'14 residuals / elevation angle



simulation residuals / elevation angle



Simulated errors

A white **measurement noise** of 1.4 mm (at 1σ) was introduced on VLBI-type measurements. Moreover standard errors were introduced on following models or parameters:

1. **Stations coordinates:** 3 cm random at 1σ per X, Y, Z coordinate

	X	Y	Z	Lat	Lon	H
12 stations - mean (m) :	0.023	0.012	-0.014	-0.017	0.002	-0.005
st. dev. (m) :	0.012	0.029	0.036	0.035	0.030	0.031

2. **Quasars coordinates:** randomly according to the standard errors of ICRF2

	Right Asc. (ms)	Decl. (mas)
60 quasars - mean :	0.002	-0.007
st. dev. :	0.010	0.081

3. **Pole coordinates:** randomly according to the standard errors of IERC04

	Xp (mas)	Yp (mas)	UT 1(ms)
over 8 weeks - mean :	0.002	0.000	-0.001
st. dev. :	0.042	0.044	0.015

4. **Troposphere models:** GPT/GMF vs. Hopfield with cut-off angle of 15 deg.

Simulation synopsis

Weekly processing over 8 consecutive weeks:

- 12 stations with 15 deg. elevation cutoff
- 60 quasars
- ~28 000 VLBI measures/week (every 200 s in average at each station)

VLBI residuals per noise type introduced:

	<i>before</i>	<i>/</i>	<i>after clock and troposphere adjustment</i>
	<i>(1st iteration)</i>		<i>(last iteration in processing)</i>
1. Stations coordinates:	37.3 mm	/	23.1 mm
2. Quasars coordinates:	2.9 mm	/	2.1 mm
3. Pole coordinates:	6.4 mm	/	4.8 mm
4. Troposphere models:	113.6 mm	/	2.0 mm
5. Measurements:	1.41 mm	/	1.37 mm
6. All effects together			24.6 mm (80 ps)

Adjusted parameters:

Troposphere zenithal bias per 2 hrs (in pwl mode) → 8064 parameters

Clock offset per 2hrs (in pwl mode) → 7392 parameters

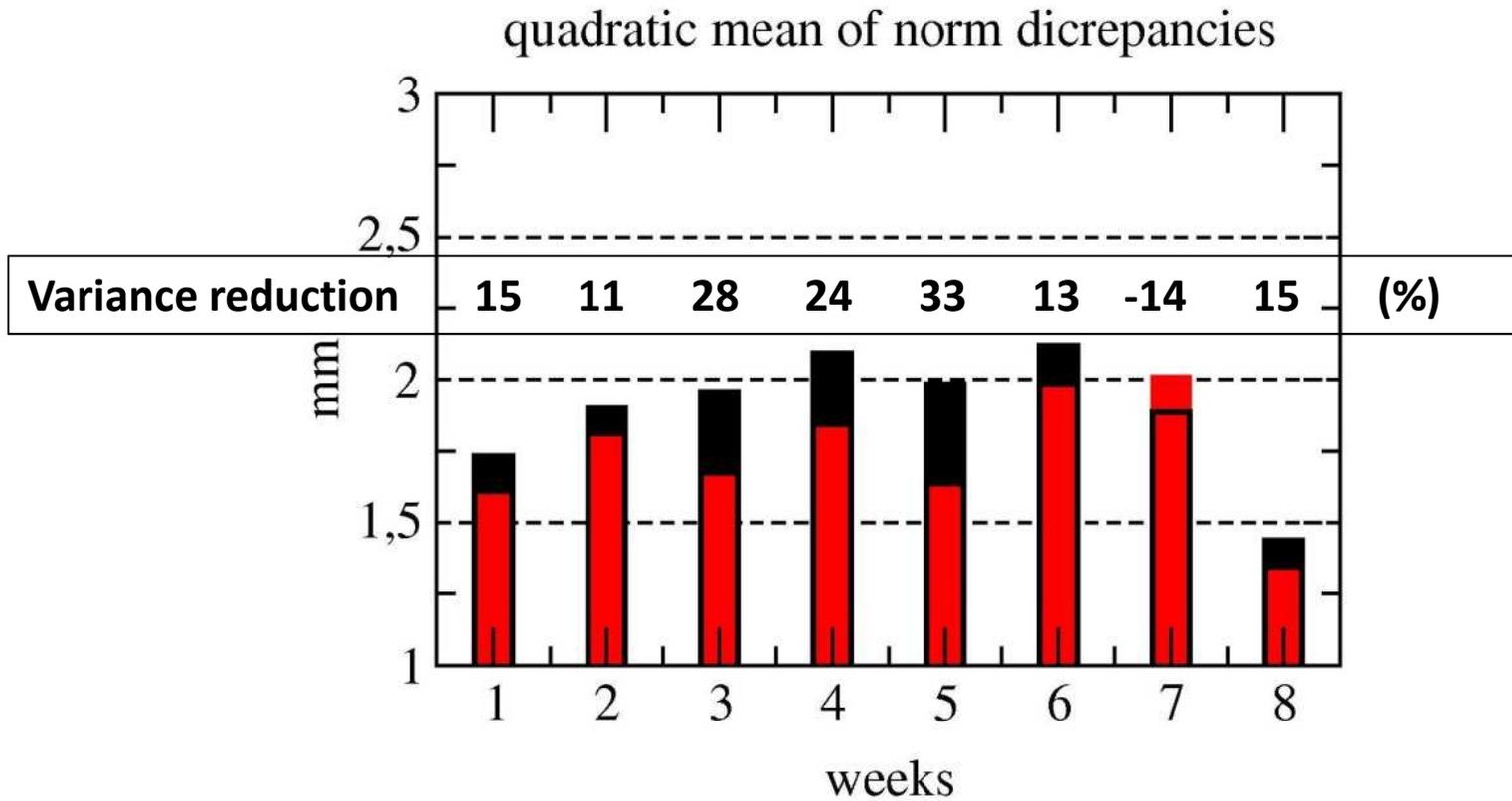
Pole coordinates per day (Px, Py, UT in pwl mode) → 171 parameters

Quasar coordinates for 60 quasars over 8 weeks (r. asc./decl.) → 120 parameters

Station coordinates for 12 stations per week (X, Y, Z) → 288 parameters

TRF results

Stations coordinates are adjusted weekly considering or not a VLBI antenna in Tahiti. One notes a general improvement of the TRF with variance reduction factors of ~15% in average after adjustment.

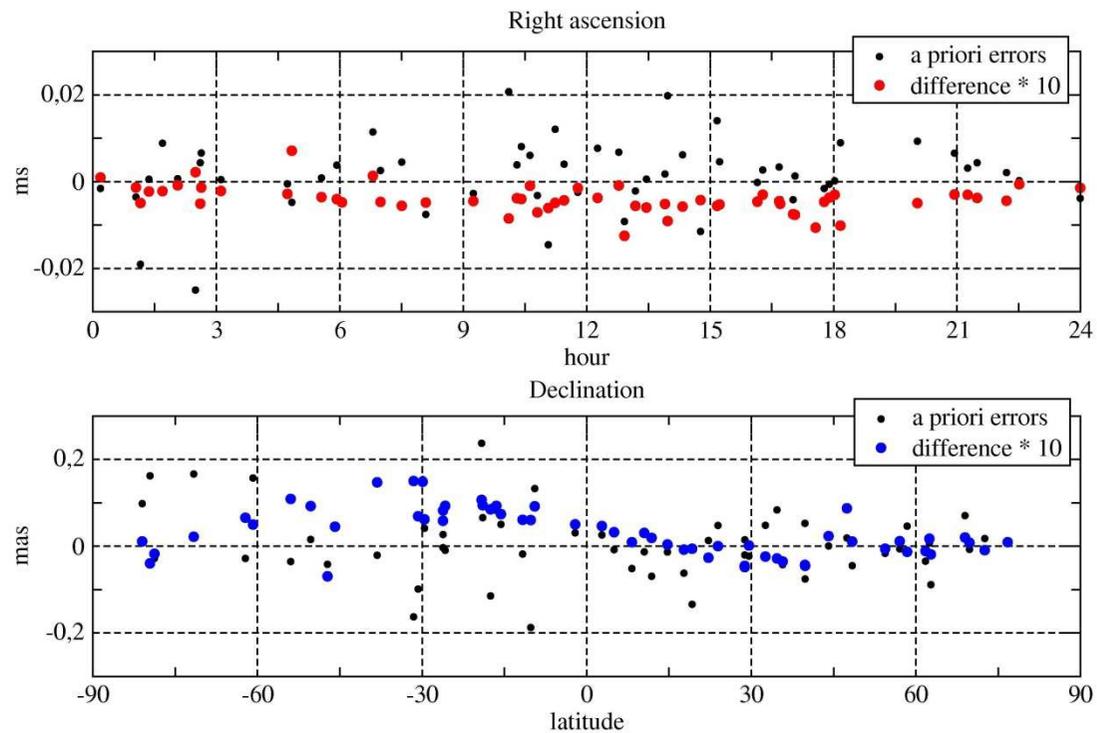


CRF results

Quasar coordinates are adjusted over the full 8-week test period with/without the Tahiti site.

rms / ref.	a priori noise	11 sta. wo T	12 sta.
R. asc. (ms)	.0105	.0022	.0025
Decl. (mas)	.0810	.0142	.0106

CRF declination is mainly improved in the southern hemisphere

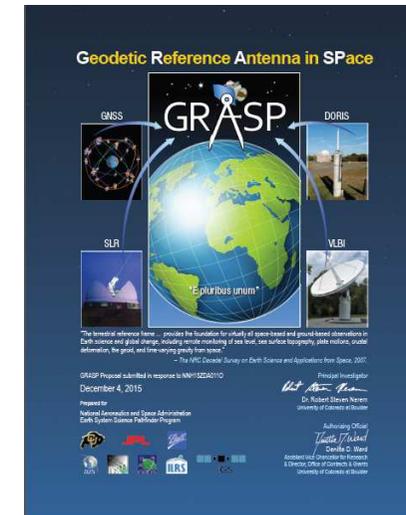


Satellite-VLBI mission projects

aiming at improving TRF to a precision of 1 mm and a stability of .1 mm/yr and homogenizing TRF/CRF/EOP

❑ GRASP

- NASA Earth Venture Mission-2 (2020)
- Payload: TriG/SLR/VLBI transmitter
- Orbit: 925 – 1400 km, sun-synchronous
- Submitted on December 4, 2015



❑ E-GRASP/Eratosthenes

- ESA Earth Explorer-9 mission (2024)
- Payload: GNSS/DORIS/SLR/VT/ μ STAR/T2L2
- Orbit: ~900 – ~7200 km, sun-synchronous
- To be submitted on June 24, 2016



ESA/EXPLORER/EE-9
November 2015

The Earth Observation Envelope Programme



Call for Proposals for Earth Explorer Mission EE-9